

SPREADING OF A VISCOUS DROP ON IMPACT

E. I. Andriankin

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The problem of the inertialess spreading of a drop due to impact compression was solved by Reynolds on the assumption of constancy of the coefficient of viscosity and was used in [1] to analyze the sensitivity of liquid explosives. It is of interest to take into account inertia forces and the variation of the coefficient of viscosity  $\mu$  with temperature since the heating of the liquid and the deceleration of the striker depend on these factors. The outcome of the solution for steady-state conditions is also discussed.

We will assume that the radius  $R$  of the base of the striker is the same as the initial radius of the thin cylindrical layer of viscous substance of thickness  $\delta_0$ . The mass of the striker is  $m$  and its initial velocity is  $V_0$ . After simplifications which depend on the smallness of the ratio  $\delta_0/R$ , the hydrodynamic equations can be written thus:

$$\begin{aligned} \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial r} + v \frac{\partial u}{\partial z} &= - \frac{1}{\rho_0} \frac{\partial p}{\partial z} + \frac{1}{\rho_0} \frac{\partial}{\partial z} \left( \mu \frac{\partial u}{\partial z} \right), \quad \frac{\partial p}{\partial z} = 0, \\ \frac{1}{r} \frac{\partial ur}{\partial r} + \frac{\partial v}{\partial z} &= 0, \quad u(r, 0, t) = 0, \quad u(r, \delta, t) = 0, \\ v(r, 0, t) = 0, \quad v(r, \delta, t) &= w, \quad p(R, t) = 0. \end{aligned} \quad (1)$$

We will assume that the liquid does not conduct heat. Then the dissipated energy in a Lagrangian particle with coordinates  $r_0$  and  $z_0$  is conserved

$$\left( \frac{dT}{dt} \right)_{r_0, z_0} = \frac{\mu}{\rho_0 c_p} \left( \frac{\partial u}{\partial z} \right)_{r, t}^2, \quad \left( \frac{\partial r}{\partial t} \right)_{r_0, z_0} = u, \quad \left( \frac{\partial z}{\partial t} \right)_{r_0, z_0} = v,$$

$$T(r_0, z_0, 0) = 0, \quad r(r_0, z_0, 0) = r_0, \quad z(r_0, z_0, 0) = z_0. \quad (2)$$

If  $R/c \ll \tau_1$  ( $c$  is the velocity of sound in the striker,  $\tau_1 = \delta_0/V_0$  is the characteristic time of impact), the deceleration of the striker can be written as:

$$m \frac{dw}{dt} = 2\pi \int_0^R pr dr, \quad \delta = \delta_0 + \int_0^t w dt, \quad w(0) = -v_0. \quad (3)$$

We solve Eq. (1) by the method of moments. We take  $u$  in the form of a series

$$u = z(\delta - z) [f_0(r, \delta) + z f_1(r, \delta) + \dots]$$

which satisfies the conditions of adhesion of the liquid at  $z = 0$  and  $z = \delta$ . Multiplying the equation of motion (1) by  $z^n$  ( $n = 0, 1, 2, \dots$ ) and integrating it over the thickness of the layer from 0 to  $\delta$  we obtain

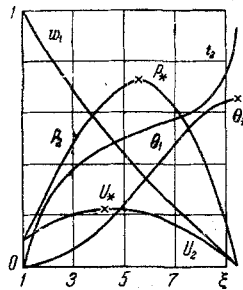


Fig. 1

tain a system of differential equations which is equivalent to (1) at the limit  $n \rightarrow \infty$ . Integrating the continuity equation with respect to  $z$  and confining ourselves to the zero approximation in Eq. (4) we find

$$v = - \frac{1}{r} \frac{\partial}{\partial r} r \int_0^z u dz, \quad \text{i.e.} \quad u = \frac{3wrz(z - \delta)}{\delta^3},$$

$$v = \frac{wz^2(3\delta - 2z)}{\delta^3}. \quad (5)$$

Using Eqs. (2) and (5) and converting from  $t$  to  $\delta$ , we establish a relationship between the moving and Lagrangian coordinates and determine the temperature distribution

$$\frac{dr}{u} = \frac{dz}{v} = \frac{d\delta}{w}, \quad \frac{d\delta}{dt} = w, \quad r = r_0 \left[ \frac{\delta_0(\delta + e)}{\delta(\delta_0 + e)} \right]^{3/4},$$

$$e = \frac{z_0 \delta_0 (\delta_0 - z_0)}{(0.5\delta_0 - z_0)^2},$$

$$z = \frac{\delta}{2} [1 \pm \sqrt{\delta / (\delta + e)}] \quad \left( \begin{array}{l} + \text{ for } z > 0.5\delta \\ - \text{ for } z < 0.5\delta \end{array} \right),$$

$$T = \frac{9}{\rho_0 c_p} \int_{\delta_0}^{\delta} \frac{\mu w r^2 (\delta - 2z)^2}{\delta^3} d\delta, \quad T_* = \frac{9\mu_0 R^2}{\rho_0 c_p} \int_0^{\delta} \frac{w}{\delta^4} d\delta. \quad (6)$$

It follows from (6) that the maximum temperature  $T_*$  is attained at the points  $r = R, z = 0$ , and  $r = R, z = \delta$ . Using Eqs. (4) and averaging Eq. (1) over  $z$ , we find

$$\frac{\partial p}{\partial r} = \frac{6w\mu_0 r}{\delta^3} (\mu_1 + \psi), \quad \psi = \frac{\rho_0 \delta^2}{12\mu_0} \frac{dw}{d\delta} - \frac{0.15\rho_0 \delta w}{\mu_0},$$

$$\mu = \mu_0 \mu_1 [p, T(r, 0, t)]. \quad (7)$$

Here  $T(r, 0, t)$  is expressed from (6) on the condition  $z = 0$  and hence,  $r = r_0$ . Integration of Eq. (7) for arbitrary variation of the viscosity with  $p$  and  $T$  requires numerical calculation even within the

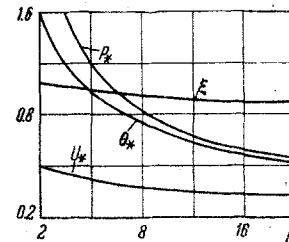


Fig. 2

framework of the assumptions made. However, in special cases the solution is obtained in quadratures

$$\int_0^p \frac{dx}{\mu_1(x) + \psi} = \frac{3w\mu_0}{\delta^3} (r^2 - R^2), \quad \mu_1 = \mu_1(p),$$

$$\int_0^{y_1} \frac{\delta^3 dx}{3w\mu_0 [A(x/\theta)^\omega + \psi] - x\delta^3} = 2 \ln \frac{r}{R},$$

$$y_1 = \frac{p}{r^2}, \quad \omega = \frac{n}{n+1},$$

$$A = \left( \frac{\rho_0 c_p T_0 \delta_0^3}{9(n+1)\mu_0 v_0} \right)^\omega, \quad \theta = \frac{T_* \rho_0 C_p \delta_0^3}{9\mu_0 R^2 v_0}, \quad \mu_1 = \left( \frac{p}{T} \right)^n,$$

$$p = \frac{3w\mu_0}{\delta^3} \left[ (r^2 - R^2) \psi + (n+1) A \theta^{-\omega} \left( r^{\frac{2}{n+1}} - R^{\frac{2}{n+1}} \right) \right],$$

$$\mu_1 = \left( \frac{T_0}{T} \right)^n. \quad (8)$$

Using Eq. (8) we obtain the equation of deceleration of the striker, which in dimensionless form for the case  $\mu_1 = (T_0/T)^n$ , for instance, is written thus:

$$(1 + \lambda\beta\xi) \frac{d}{d\xi} \xi^{-2} \frac{d\theta}{d\xi} + 1.8\lambda\beta\xi^{-2} \frac{d\theta}{d\xi} + \lambda\kappa\xi\theta^{-\alpha} = 0,$$

$$\theta = \int_0^{\xi} \xi^2 w_1 d\xi,$$

$$\theta(1) = 0, \quad (d\theta/d\xi)_1 = 1, \quad w = -w_1 v_0,$$

$$\xi = \delta_0 / \delta, \quad \beta = 0.15\rho_0 v_0 \delta_0 / \mu_0,$$

$$\lambda = 5\pi\mu_0 R^4 / 6m v_0 \delta_0^2,$$

$$\kappa = 3.6(n+1)AR^{-2n} / (n+2). \quad (9)$$

This equation is easily integrated numerically. If  $\beta \rightarrow 0$ , then Eqs. (9) allows a similarity group and on substitution of the variables  $v = \theta\xi^\epsilon$ ,  $z = dv/d\ln\xi$ ,  $\epsilon = -5(n+1)/(2n+1)$  reduces to an equation of the first order. If  $n = 0$ , i. e.,  $\mu = \text{const}$ , then Eq. (9) is integrated in finite form

$$w_1 = \left(\frac{y_0}{y}\right)^{1.8} + \frac{1}{\lambda\beta^2} \left[1 - \left(\frac{y_0}{y}\right)^{1.8}\right] - \frac{9y}{14\lambda\beta^2} \left[1 - \left(\frac{y_0}{y}\right)^{2.8}\right],$$

$$\tau_1 = \int_1^{\xi} w_1^{-1} \xi^{-2} d\xi,$$

$$y = 1 + \lambda\beta\xi, \quad y_0 = 1 + \lambda\beta, \quad t_1 = t / \tau_1. \quad (10)$$

It follows from (10) that at large  $\xi$  we have  $w_1 \approx 1/\lambda\beta^2 - 9\xi/14\beta$ . Hence, the striker stops when the layer has a finite thickness  $\xi = \xi_*$ .

Close to  $\xi = \xi_*$  we have  $w_1 = 18\lambda\xi_*(\xi_* - \xi)/(1 + \lambda\beta\xi_*)$  and, hence, when  $\xi \rightarrow \xi_*$ , the value of  $t_1$  increases as  $\ln[1/(\xi_* - \xi)]$ .

At low Reynolds numbers the inertia forces become smaller than the viscosity forces and relationships (10) give the law of motion of the striker in the problem of [1]

$$w_1 = 1 + 0.9\lambda(1 - \xi^2),$$

$$0.9\lambda t_1 = \frac{1}{a^2} \left[1 - \frac{1}{\xi} + \frac{1}{2a} \ln \frac{(a + \xi)(a - 1)}{(a - \xi)(a + 1)}\right],$$

$$a^2 = \frac{1 + 0.9\lambda}{0.9\lambda}. \quad (11)$$

It follows from (10) that the velocity of the striker decreases monotonically with time. However, the radial velocity of flow at the point  $z_1 = 0.5\delta$ ,  $r = R$  at instant  $\xi_1$  and pressure at point  $r = 0$  for  $\xi = \xi_2$  attain maxima  $u_1(\xi_1)$  and  $p_1(\xi_2)$ , which are determined from the relationships

$$u_1 = \frac{4\delta_0 u(R, z_1, \xi)}{3v_0 R} = \xi w_1,$$

$$p_1 = \frac{\delta_0^3 p(0, \xi)}{3\mu_0 v_0 R^2} = w_1 \xi^2 \frac{\beta w_1 + \xi}{1 + \lambda\beta\xi}, \quad (12)$$

$$w_0(1 - 0.8\lambda\beta\xi_1) = 1.8\mu\xi_1^2,$$

$$2\beta(1 - 0.8\lambda\beta\xi_2)w_2^2 + 3\xi_2(1 - 0.8\lambda\beta\xi_2)w_2 = 1.8\lambda\xi_2^3,$$

$$w_0 = w_1(\xi_1), \quad w_2 = w_1(\xi_2). \quad (13)$$

When  $\beta \rightarrow 0$  we find from Eqs. (11) and (13) that  $\xi_1 = a^{1/5} 0.5$ ,  $\xi_2 = a^{3/5} 0.5$ . Figure 1 gives the results of calculations of the dimensionless pressure  $p_2 = 0.01 p_1$ , the temperature  $\theta_1 = 0.01 \theta$ , the time  $t_2 = 0.5t_1$ , and the velocities of the liquid  $u_2 = 0.1u_1$  and striker  $w_1$  in relation to  $\xi$  for  $\beta = 10$  and  $\lambda = 0.01$ .

In Fig. 2 these quantities at the maximum points are plotted as functions of  $\beta$ :  $u_* = u_2(\xi_1)$ ,  $p_* = p_2(\xi_1)$ ,  $\theta_1 = \theta(\xi_1)$  and  $\xi_0 = 0.1\xi_1$ .

The graphs show that the maximum velocity of outward flow is several times greater than the initial velocity and the maximum pressure (at small  $\delta$ ) is even two orders greater. If the energy of elastic

strain of the striker is greater than  $mv_0^2/2$ , the compressibility of the striker must be taken into account. Since the compression time  $t_1$  increases logarithmically, the characteristic time is of the order of unity.

We can consider the problem of the spread of a drop between two infinite plates. In this case the variable radius of the drop is expressed from the law of conservation of mass  $R^2(t)\delta = R_0^2\delta_0$  and the problem is solved in the same way as above. However, such a procedure is valid only in cases where the cumulative splash of the liquid at  $z = \delta/2$  can be neglected.

We will now estimate the correctness of the hypothesis of a quasi-stationary parabolic velocity profile in the Reynolds problem ( $\beta \rightarrow 0$ ) in the case of impact compression. The hypothesis of quasistationarity is valid if the characteristic time of formation of the viscous velocity profile  $\tau_2 \ll \tau_1$ . If we consider the dimensions, then  $\tau_2 \sim \rho_0 \delta_0^2 / \mu_0$ . However, the solution of the simplified problem of development of a viscous flow shows that  $\tau_2$  is an order less. This is important, since the solution depends exponentially on the time.

We will consider an instant  $t_0$ , close to the initial instant, but such that the pressure over the thickness of the layer manages to even out:  $\tau_0 > \delta_0/c$ . We will assume the velocity of the striker to be constant and we will take the initial distribution of  $u$  in the form  $u(r, z, 0) = v_0 r / 2\delta_0$ , which satisfies the continuity equation. In the equations of motion (1) we neglect the inertial terms, and retain the derivative  $\partial u / \partial t$ . The solution of Eqs. (17) shows that this is valid everywhere, except for the layer near the wall, where when  $t \rightarrow 0$  the velocity gradients  $\sim t^{-0.5}$  are of the same order as the acceleration [2]. However, physical sense demands that  $t > t_0$ . To get rid of the "movable boundary"  $\delta(t)$ , we convert to new variables  $\eta = z/\delta$  and  $\tau = (\xi - 1)/\delta_0$ . Then the system of equations, like (1), is written thus:

$$\rho_0 v_0 \frac{\partial u}{\partial \tau} + \rho_0 v_0 \delta \frac{\partial u}{\partial \eta} + \frac{\partial \varphi}{\partial r} = \mu_0 \frac{\partial^2 u}{\partial \eta^2},$$

$$\int_0^1 u d\eta = \frac{v_0 r}{2} \left(\tau + \frac{1}{\delta_0}\right),$$

$$\varphi = p\delta^2, \quad u(r, \eta, 0) = v_0 r / 2\delta_0, \quad u(r, 0, \tau) = 0,$$

$$u(r, 1, \tau) = 0, \quad \varphi(R, \tau) = 0. \quad (14)$$

We omit the term with  $\delta$  in (14), since it is of the order  $v\delta u/\partial y$  of the discarded inertia terms. Performing the Laplace transformation we obtain equations for the velocity images  $u \neq U$  and the pressure analog  $\varphi \neq P$ .

$$\rho_0 v_0 \delta U - \frac{\rho_0 v_0^3}{2\delta_0} r + \frac{\partial P}{\partial r} = \mu_0 \frac{\partial^2 U}{\partial \eta^2},$$

$$\int_0^1 U d\eta = \frac{v_0 r}{2s} \left(\frac{1}{s} + \frac{1}{\delta_0}\right),$$

$$U(r, 0) = 0, \quad U(r, 1) = 0, \quad P(R) = 0. \quad (15)$$

Integrating Eq. (15), we obtain

$$U = \frac{v_0 r (s + \delta_0) x \operatorname{sh} x \eta \operatorname{sh} x (1 - \eta)}{\delta_0 s^2 (x - \operatorname{th} x) \operatorname{ch} x},$$

$$P = \frac{\rho_0 v_0^3 (R^2 - r^2)}{4\delta_0} \left[ \frac{(s + \delta_0) x}{s(x - \operatorname{th} x)} - 1 \right],$$

$$x = 0.5 \left( \frac{\rho_0 v_0 s}{\mu_0} \right)^{1/2}. \quad (16)$$

It is easy to find the asymptotic form of Eq. (16) when  $s \rightarrow \infty$

$$U = \frac{v_0 r}{2\delta_0 s} [1 + 1/x -$$

$$- \exp(-2x\eta) - \exp[-2x(1-\eta)] + \dots],$$

$$P = \frac{\rho_0 v_0^3}{4} \left[ \frac{1}{s} + \frac{1}{x\delta_0} \right] (R^2 - r^2), \quad \text{for } t \rightarrow 0,$$

$$u = \frac{v_0 r_0}{2\delta_0} \left[ 1 + 4 \left( \frac{\mu_0 \tau}{\pi \rho_0 v_0} \right)^{1/2} - \operatorname{Erfi} \frac{\eta}{2} \left( \frac{\rho_0 v_0}{\mu_0 \tau} \right)^{1/2} - \right.$$

$$- \operatorname{Erf} \frac{1-\eta}{2} \left( \frac{\sigma_0 v_0}{\mu_0 \tau} \right)^{1/2} + \dots \Big],$$

$$p = \frac{\rho_0 v_0^2}{4\delta^2} (R^2 - r^2) \left[ \frac{2}{\delta_0} \left( \frac{\mu_0}{\pi \rho_0 v_0 \tau} \right)^{1/2} \exp(-\tau) + 1 + \dots \right]. \quad (17)$$

It follows from (17) that  $p$  increases infinitely when  $t \rightarrow 0$ . This is due to the fact that instantaneous stoppage of the wall layer of liquid requires infinite friction forces  $\sim t^{-0.5}$ , counterbalanced by inertial forces and the pressure, which is assumed constant over the thickness of the layer. The formulas for the originals for  $t > 0$  can be obtained from Eqs. (16) by expanding  $U$  and  $\varphi$  in power series and using the residue theorem [3]. Special points in the plane  $s$  lie on the left of the imaginary axis. At point  $s = 0$  there is a pole of the second order. Simple poles are situated at the points  $s_k = -4\mu\lambda_k^2 / \rho_0 v_0$ ,  $\operatorname{tg} \alpha_k = \lambda_k (\lambda_1 = 4.493)$  and, hence,

$$u = \frac{3v_0 r \eta (1-\eta)}{\delta} +$$

$$+ \sum_{k=1}^{\infty} \frac{2v_0 r (s_k + \delta_0) \operatorname{sh} \eta x_k \operatorname{sh} (1-\eta) x_k}{\delta_0 s_k x_k \operatorname{sh} x_k} \exp \left( -\frac{4\mu\lambda_k^2 \tau}{\rho_0 v_0} \right).$$

$$p = \frac{3v_0 \mu_0 (R^2 - r^2)}{\delta^3} \times$$

$$\times \left[ 1 + 0.1 \frac{\rho_0 v_0 \delta}{\mu} + \frac{2}{3} \frac{\delta}{\delta_0} \sum \frac{\delta_0 + s_k}{s_k} \exp \left( -\frac{4\mu\lambda_k^2 \tau}{\rho_0 v_0} \right) \right]. \quad (18)$$

It is clear from (18) that the solution rapidly (with characteristic time  $\tau_3 = \tau_2 / 4\lambda_1$ ) reaches the quasistationary regime.

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Institute of Chemical Physics, AS USSR